

# 4 – Force Detection

ME-426 – Micro/Nanomechanical Devices

Prof. Guillermo Villanueva

# EPFL Schedule

Week	Day	Theory			Question by
1	19.09	Introduction; Scaling laws, Fabrication*			
2	26.09	Transduction of the motion – Techniques			
3	03.10	Transduction of the motion – Techniques		Paper (1)	30.09.2022
4	10.10			Paper (1)	
5	17.10	Static deflection – Force, Linear behavior, Intro noise			
6	24.10	Station deflection – Stress		Paper (2)	21.10.2022
7	31.10	Dynamic behavior – Linear Analysis		Paper (3)	28.10.2022
8	08.11	Dynamic behavior – Stress influence on frequency			
9	14.11	Dynamic behavior – Mode coupling Dynamic behavior – Nonlinearities, Parametric drive			
10	21.11	Material properties – Size effects		Paper (4)	18.11.2022
11	28.11			Paper (5,6)	25.11.2022
12	05.12	Quality factor – Energy losses Thermomechanical noise, Detection schemes			
13	12.12			Paper (7)	09.12.2022
14	19.12			Paper (8, 9)	09.12.2022

ME426 - Schedule

2

# EPFL Practical matters – Let's connect

- Smartphones

- iPhone
- Android
- Blackberry

} Download and install the app "*PointSolutions*"



TurningPoint

- Computers (or Windows Phones)

- Go to [responseware.eu](http://responseware.eu)

- Choose EUROPE Server (if asked)

- Join session – ME426

- When asked – do not sign in, do not input your name. Just enter as guest.

# EPFL Introduction

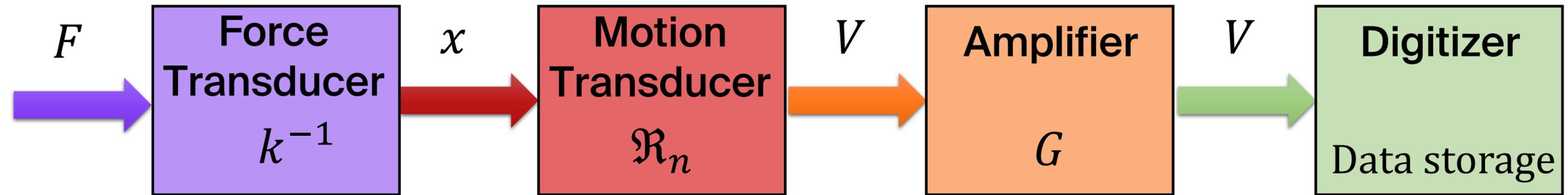
- Force detection
- Transduction chain
- Linear behavior
- Lumped element model
- Nonlinear behavior
- Composite beams
- Accelerometers

**EPFL**



**Force detection**

# EPFL Force detection

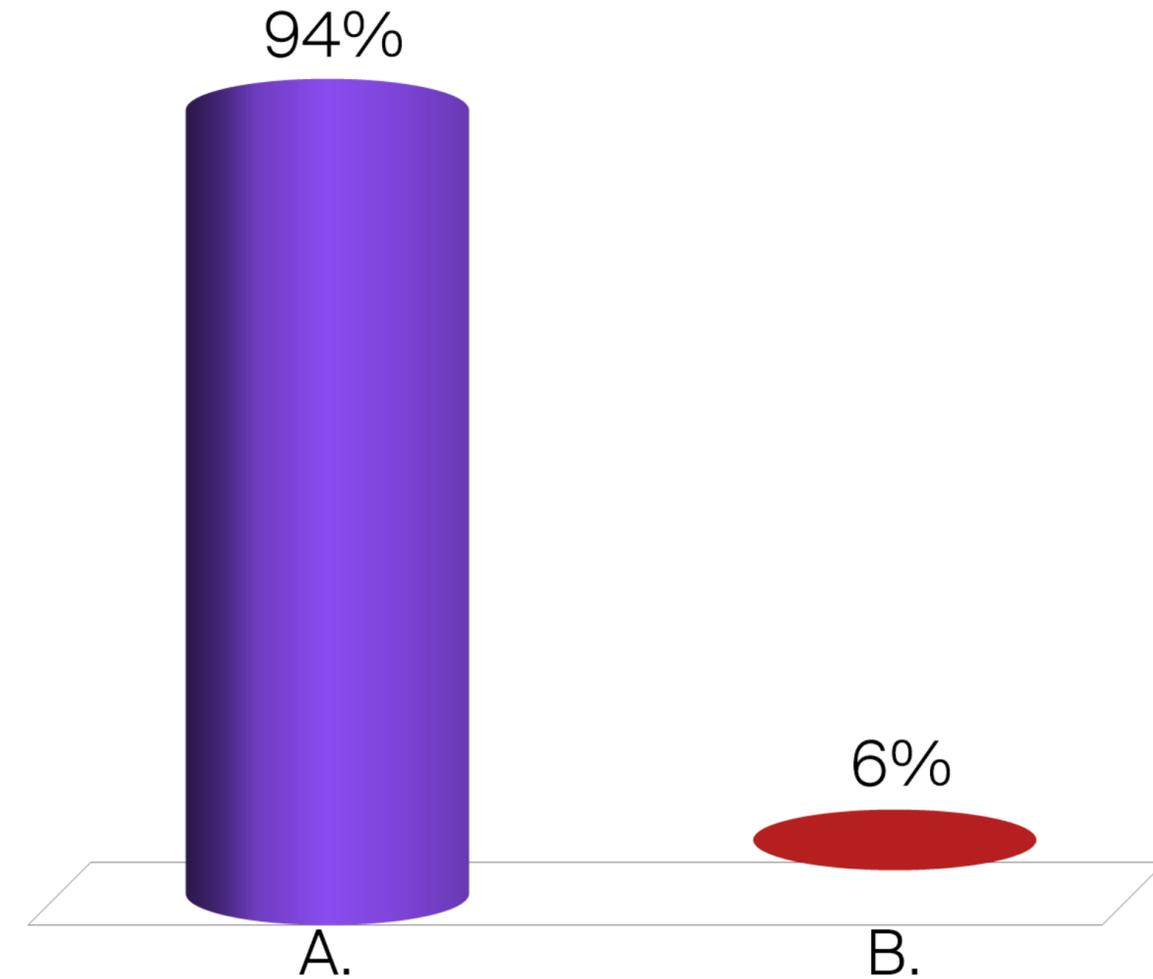


# EPFL Which structure is better to detect a force?

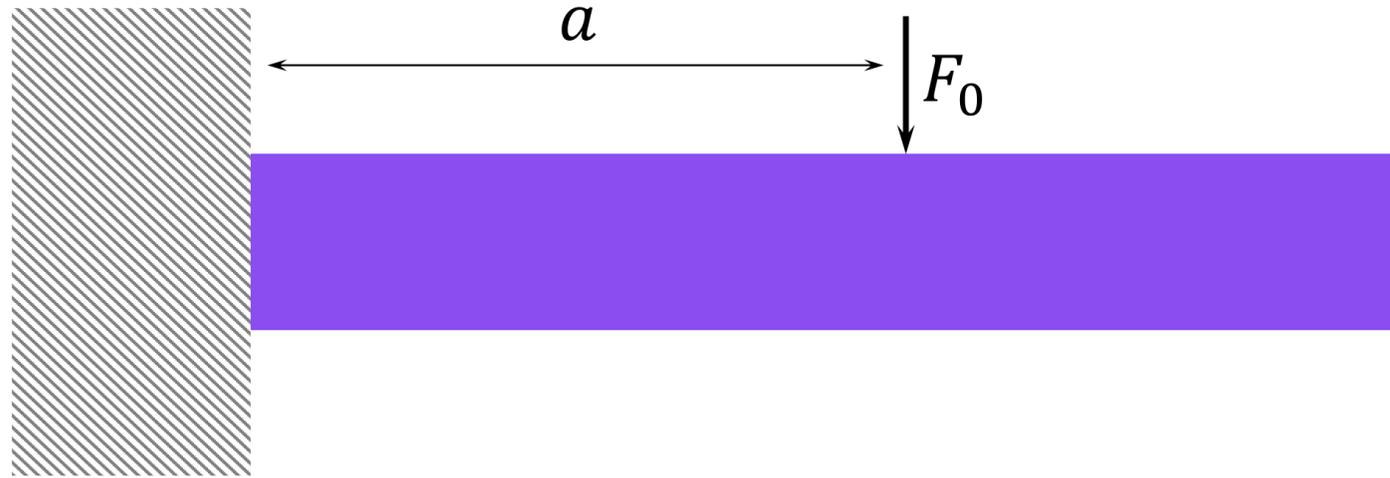


A. Flexural beam

B. Elongational beam

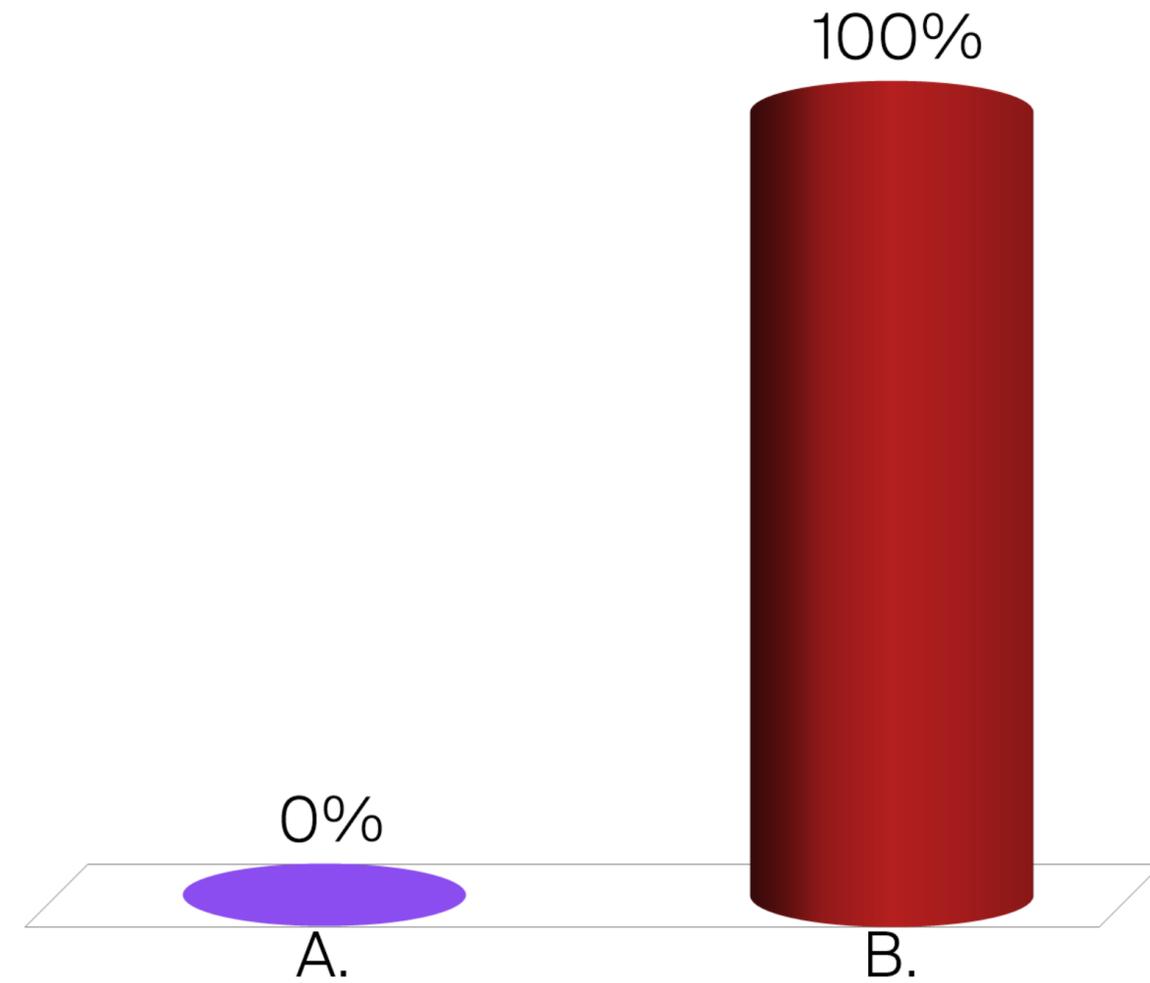


# EPFL Is the minimum detectable force the same for all $a$ ?

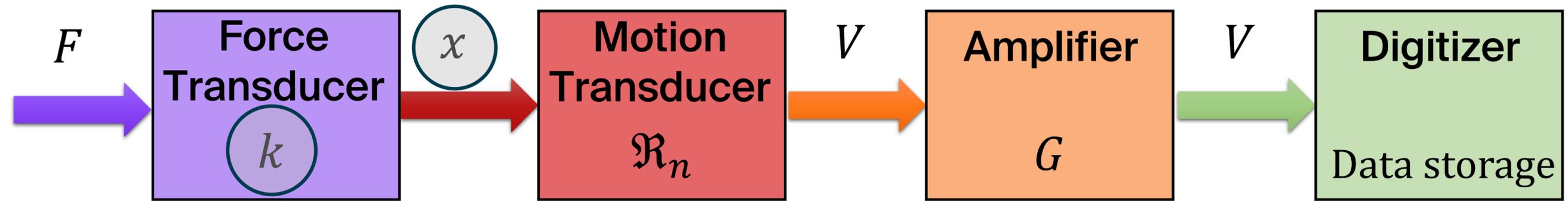


A. Yes

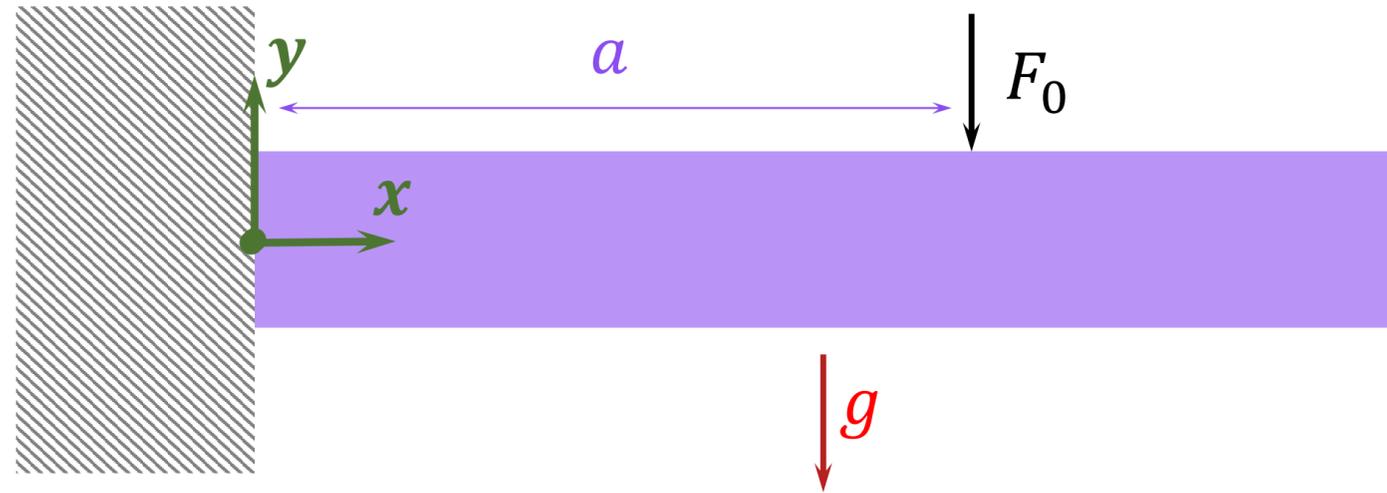
B. No



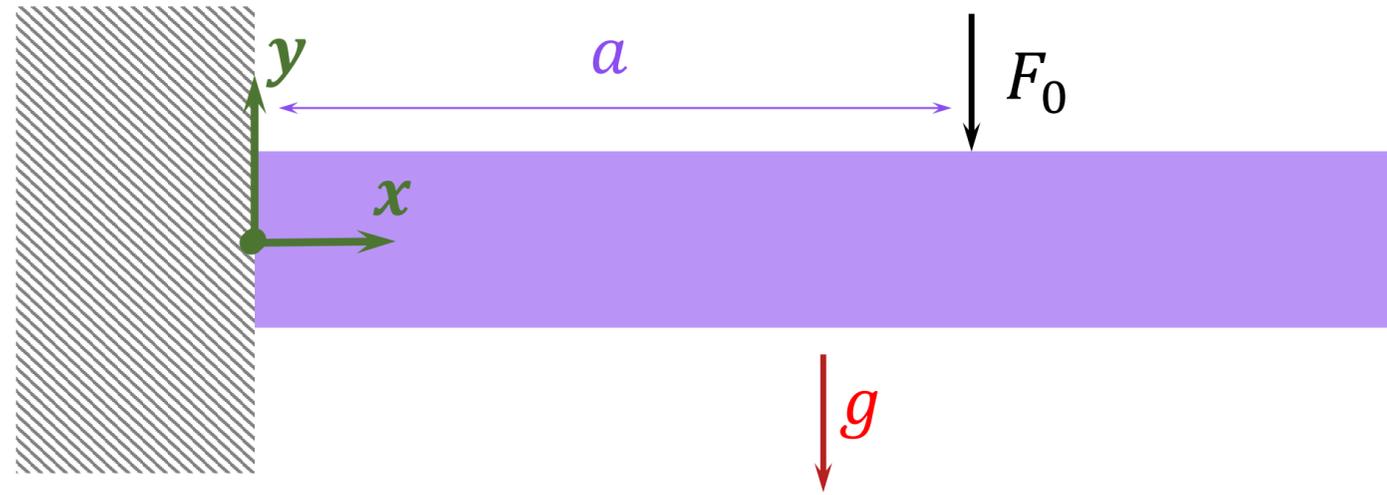
# EPFL Force detection



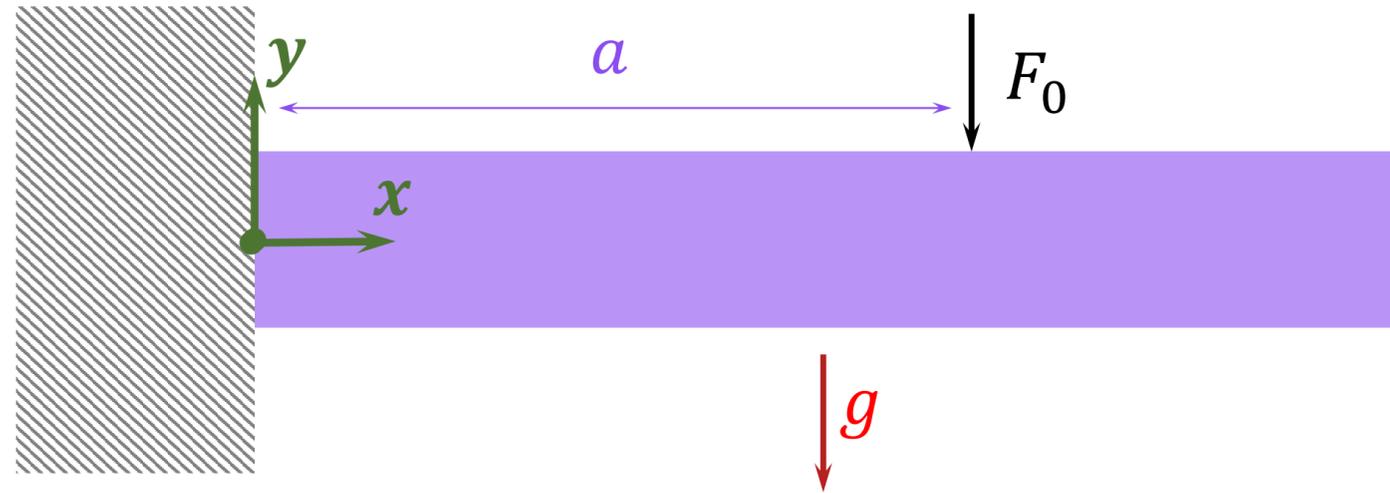
# EPFL Let's do some basics



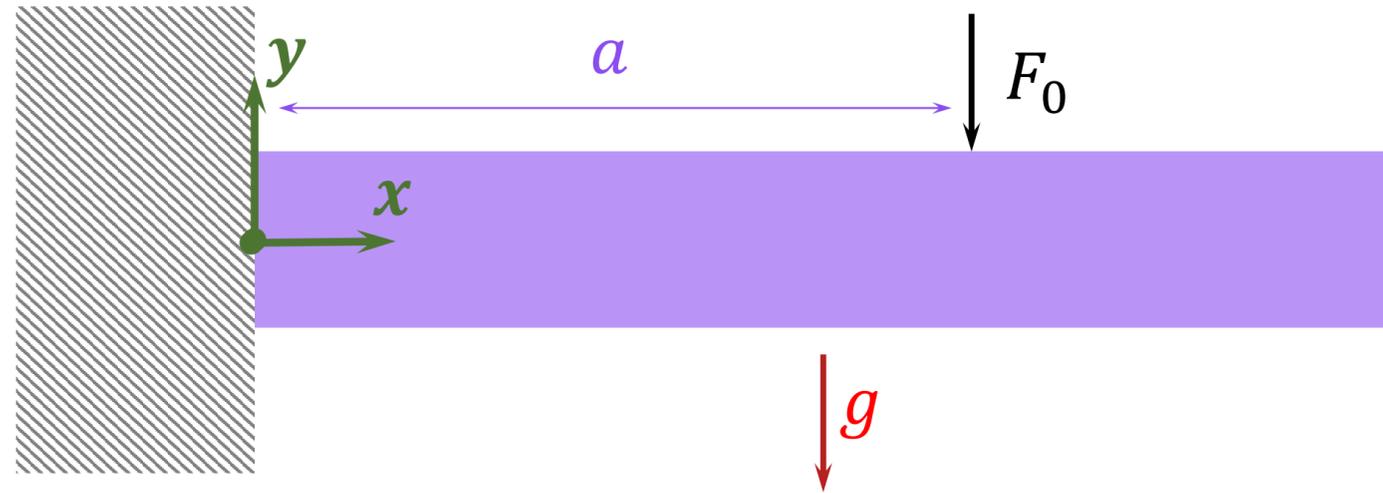
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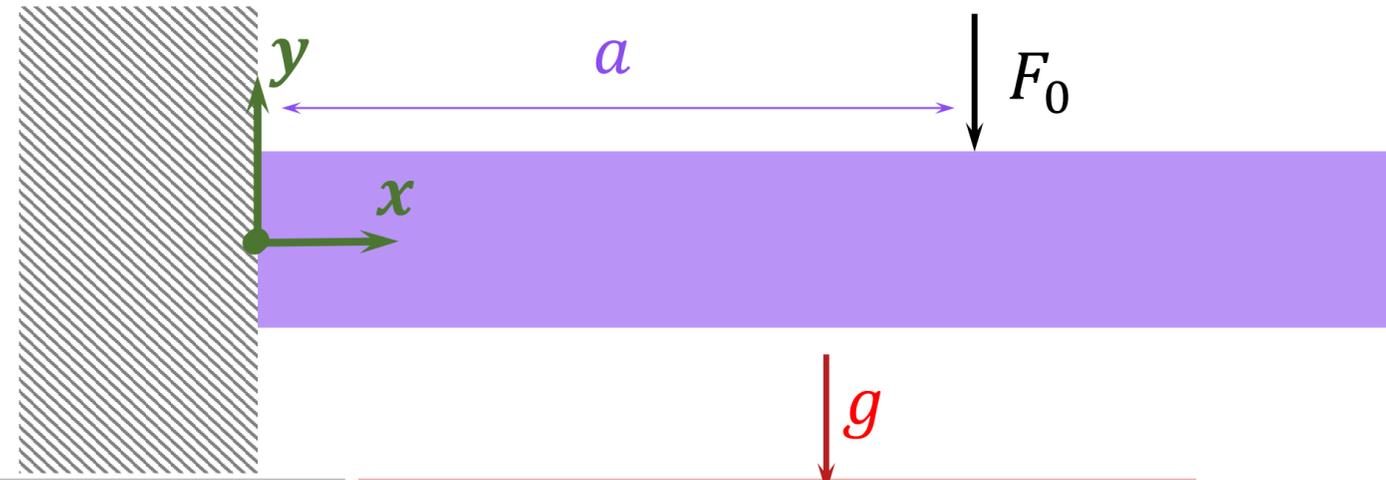
# EPFL Let's do some basics



$$V(x) = \begin{cases} F_0 + \rho g A(L - x) & \text{if } x < a \\ \rho g A(L - x) & \text{if } x > a \end{cases} \quad M(x) = \begin{cases} F_0(x - a) - \frac{\rho g A}{2}(x - L)^2 & \text{if } x \leq a \\ -\frac{\rho g A}{2}(x - L)^2 & \text{if } x \geq a \end{cases}$$

$$w(x) = \begin{cases} \frac{F_0}{6EI} x^2(x - 3a) - \frac{\rho g A}{12EI} x^2(6L^2 - 4Lx + x^2) & \text{if } x \leq a \\ \frac{F_0}{6EI} a(-2a^2 - 3xa) - \frac{\rho g A}{12EI} x^2(6L^2 - 4Lx + x^2) & \text{if } x \geq a \end{cases}$$

# EPFL Let's put some numbers

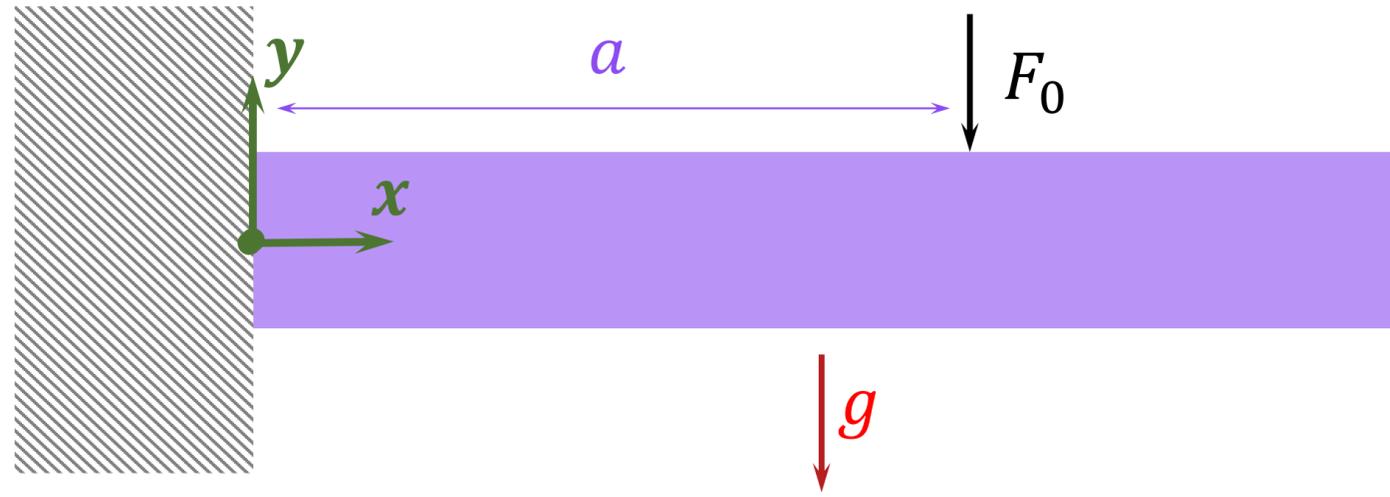


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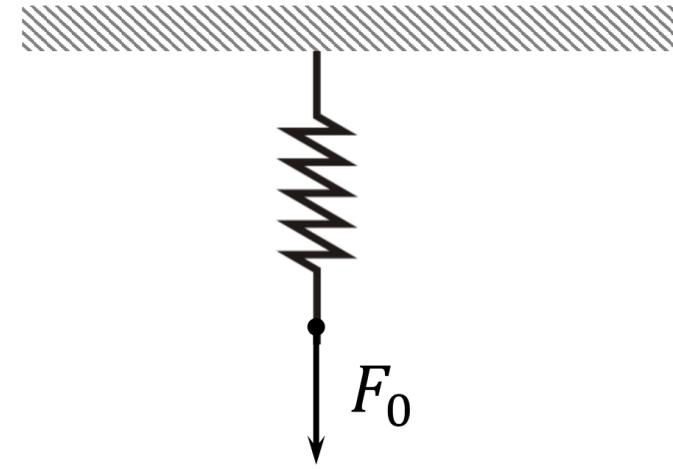
Device	Dimensions (m)	Max. deflection gravity (m)	Force to match gravity (N)
Diving board	$10 \times 1 \times 0.1$	0.3	$25 \cdot 10^3$
Tuning fork	$0.1 \times 10^{-3} \times 10^{-3}$	$3 \cdot 10^{-5}$	$2 \cdot 10^{-3}$
Microcantilever	$(500 \times 50 \times 5) \cdot 10^{-6}$	$10^{-9}$	$4 \cdot 10^{-9}$
Nanocantilever	$(10 \times 0.5 \times 0.1) \cdot 10^{-6}$	$3 \cdot 10^{-13}$	$10^{-14}$

In micro/nano scale, gravity and weight can be neglected – but maybe important for vibrations

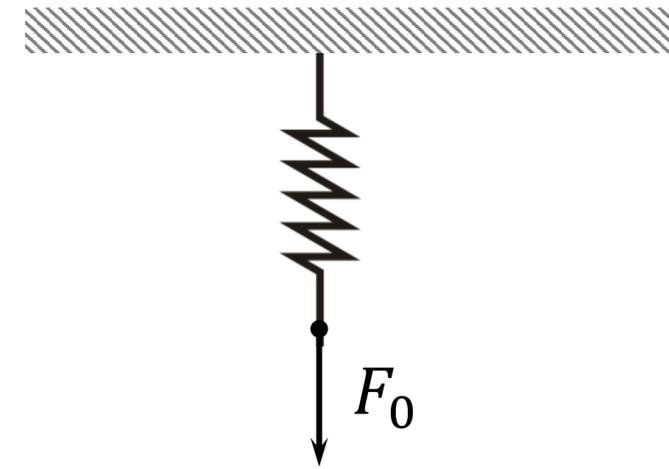
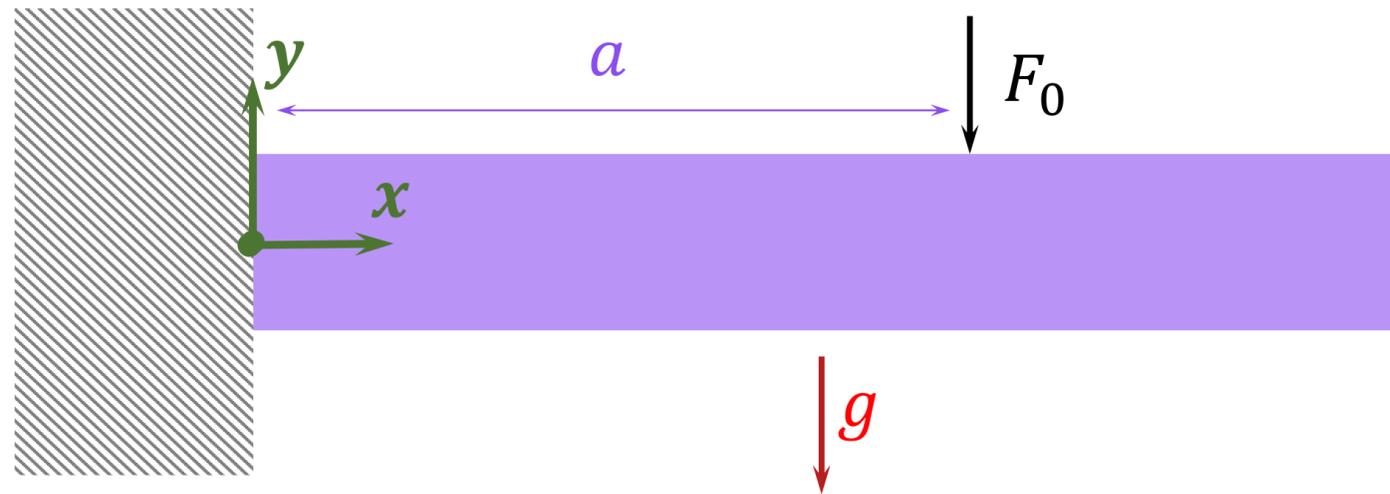
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$$k = \frac{F_0}{w(x)}$$

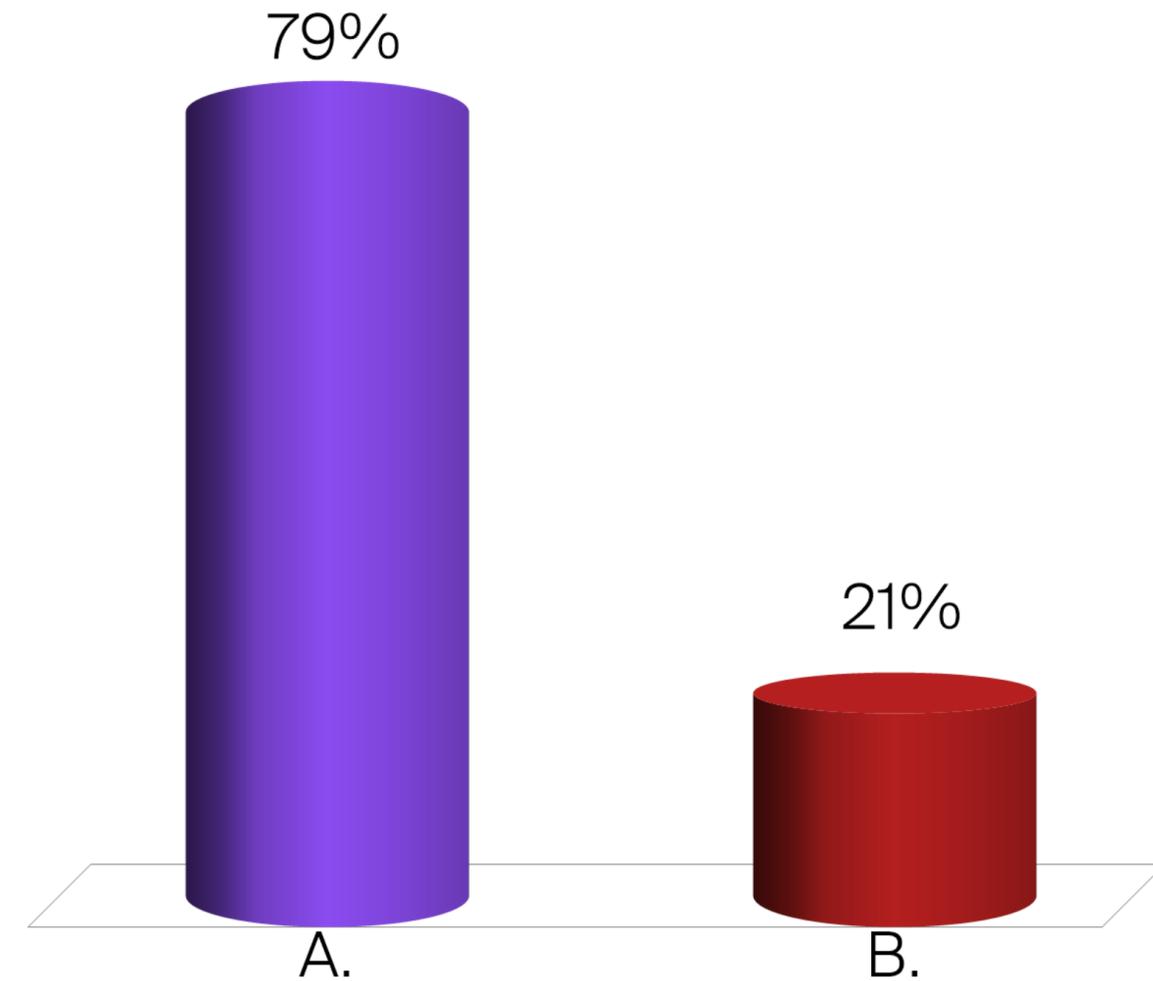
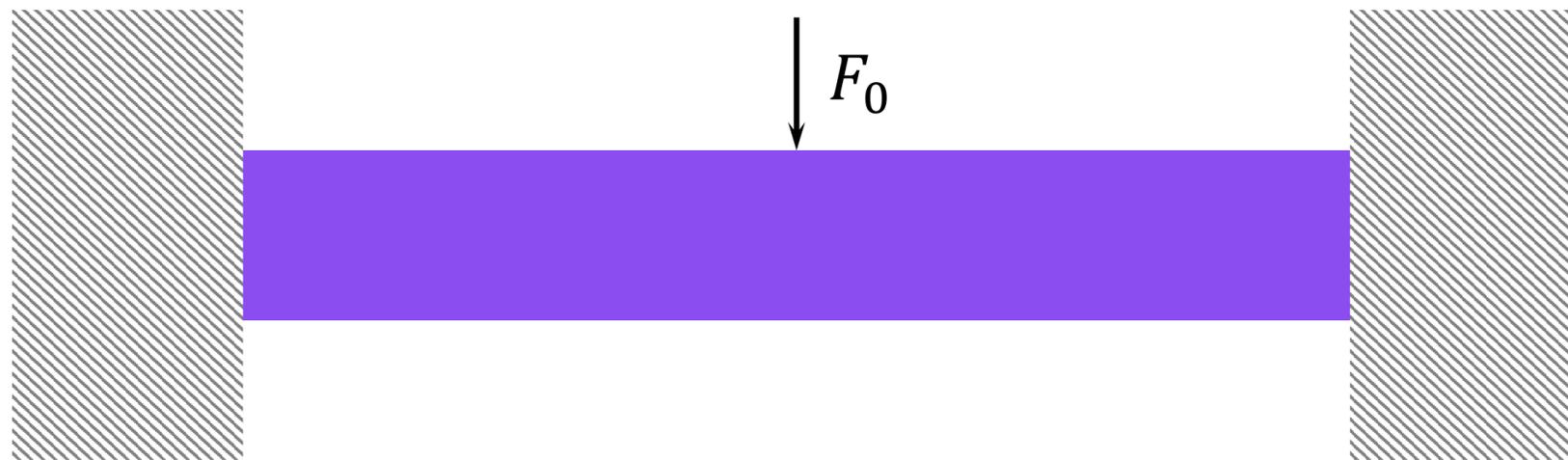
- Definition of elastic constant seems easy enough
- But what displacement should we take?
- Is the  $k$  different depending on where we apply the force?

# EPFL Which structure is better to detect a force?

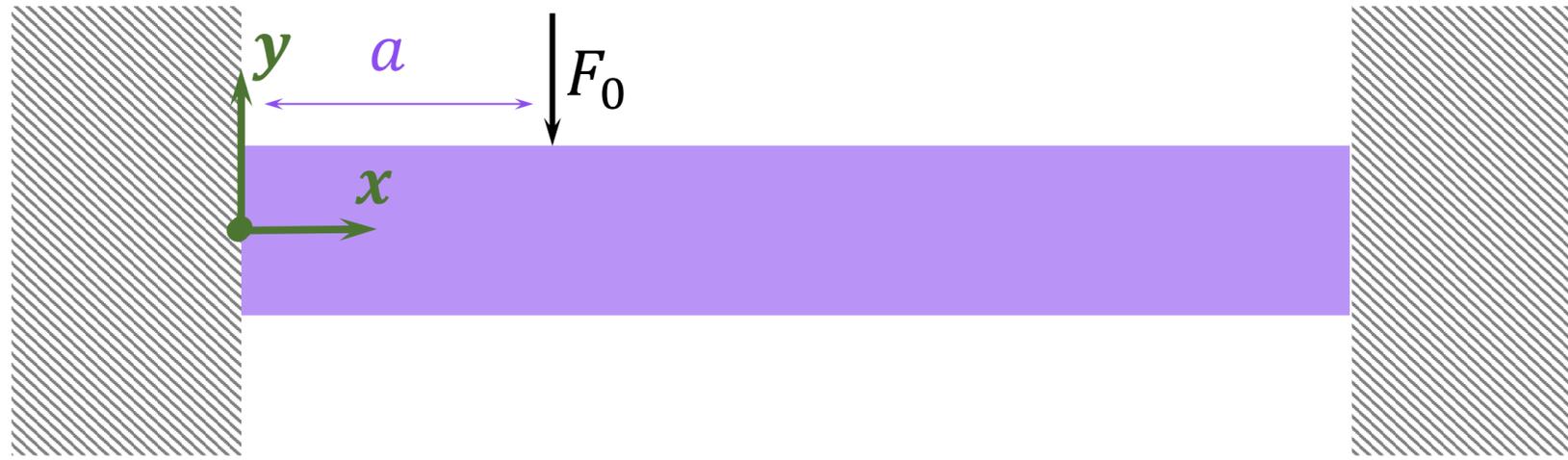


A. Cantilever beam

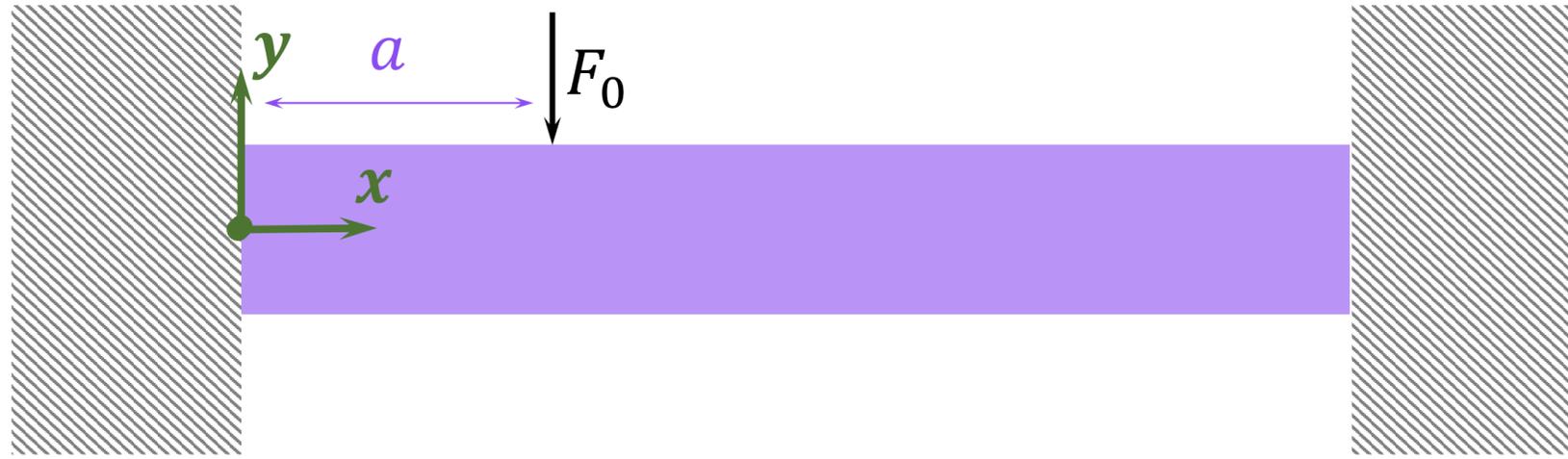
B. Clamped-Clamped beam



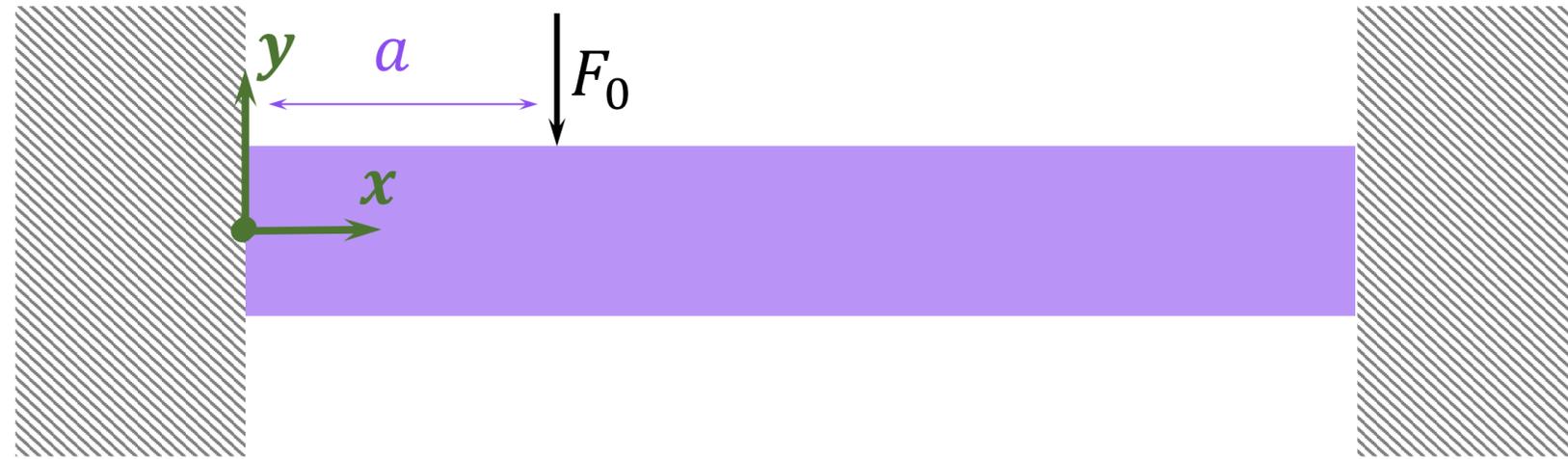
# EPFL Clamped-Clamped beam



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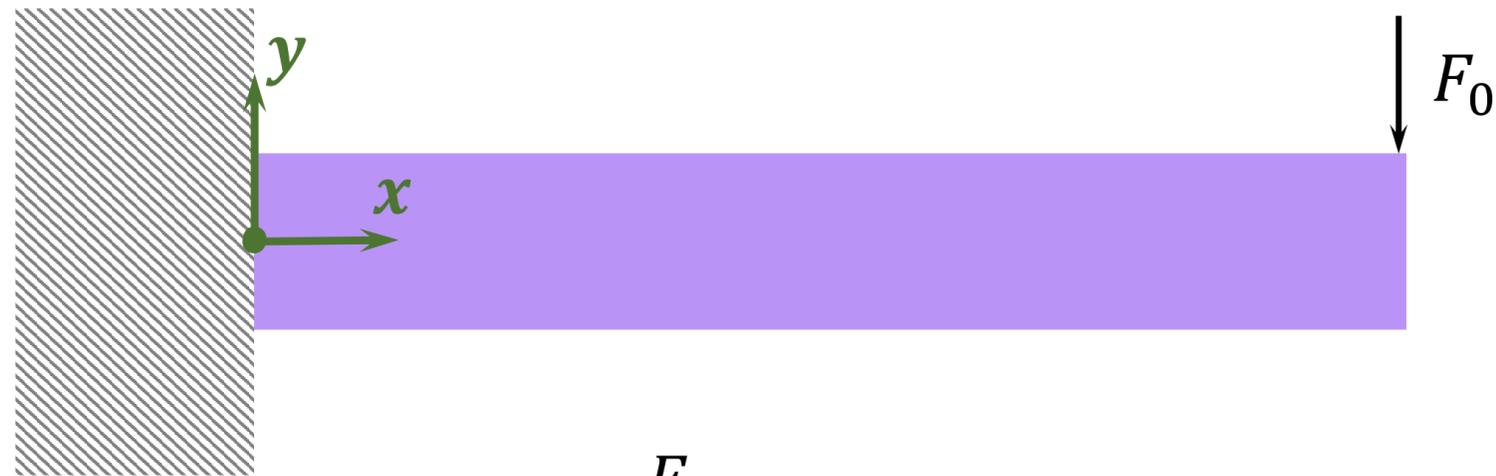


$$w(x) = \begin{cases} \frac{F_0}{6EI} x^2 \left(1 - \frac{a}{L}\right)^2 \left(-3a + x \left(1 + \frac{2a}{L}\right)\right) & \text{if } x \leq a \\ \frac{F_0}{6EI} a^2 \left(1 - \frac{x}{L}\right)^2 \left(-3x + a \left(1 + \frac{2x}{L}\right)\right) & \text{if } x \geq a \end{cases}$$

- Definition of elastic constant seems easy enough
- But what displacement should we take?
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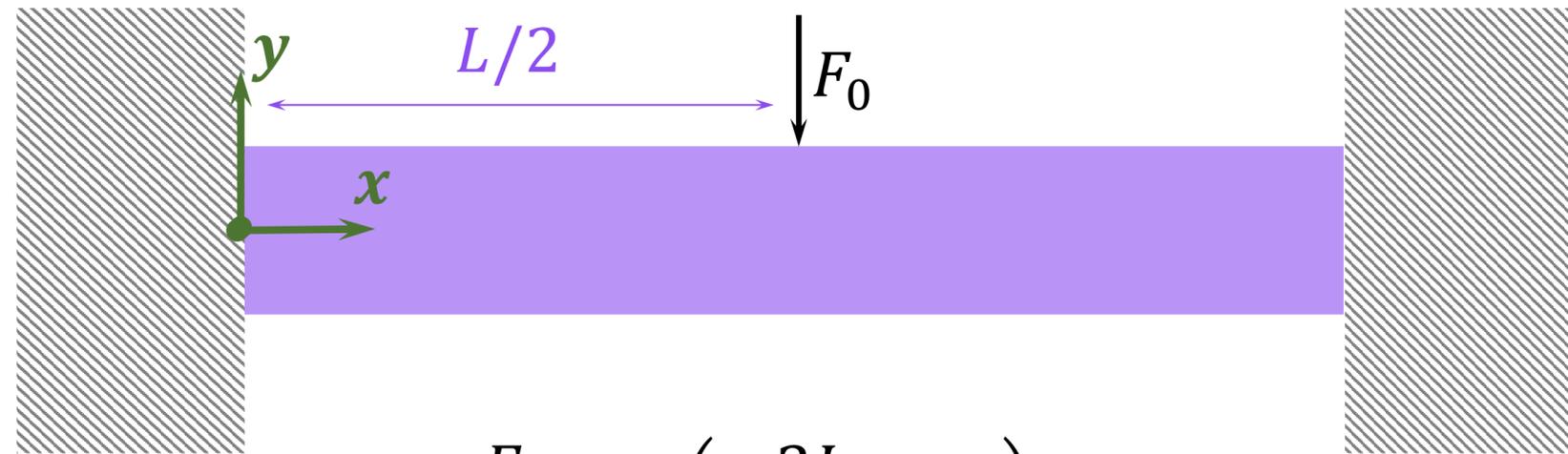
$$k = \frac{F_0}{w(x)}$$

# EPFL Comparing cantilever and C-C beam



$$w(x) = \frac{F_0}{6EI} x^2 (x - 3L)$$

$$w(L) = -\frac{F_0}{3EI} L^3$$



$$w(x) = \frac{F_0}{24EI} x^2 \left( -\frac{3L}{2} + 2x \right) \quad \text{if } x \leq L/2$$

$$w\left(\frac{L}{2}\right) = -\frac{F_0}{192EI} L^3$$

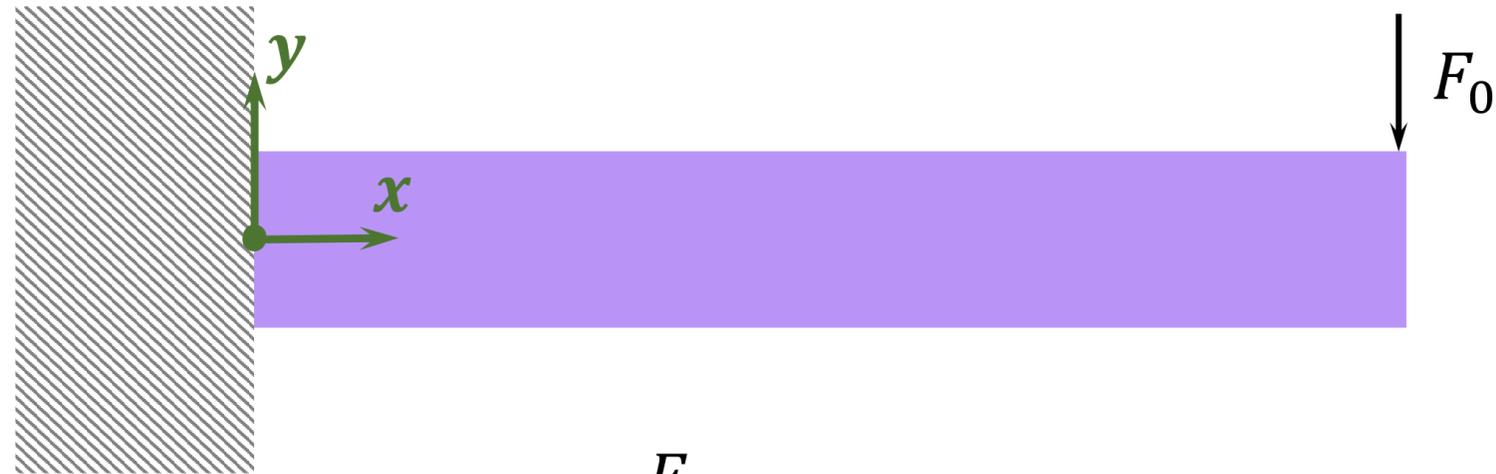
Dimensions (m)	Cantilever deflection to 1 $\mu\text{N}$ (m)	C-C beam deflection to 1 $\mu\text{N}$ (m)
$10 \times 1 \times 0.1$	$10^{-11}$	$10^{-13}$
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**EPFL**



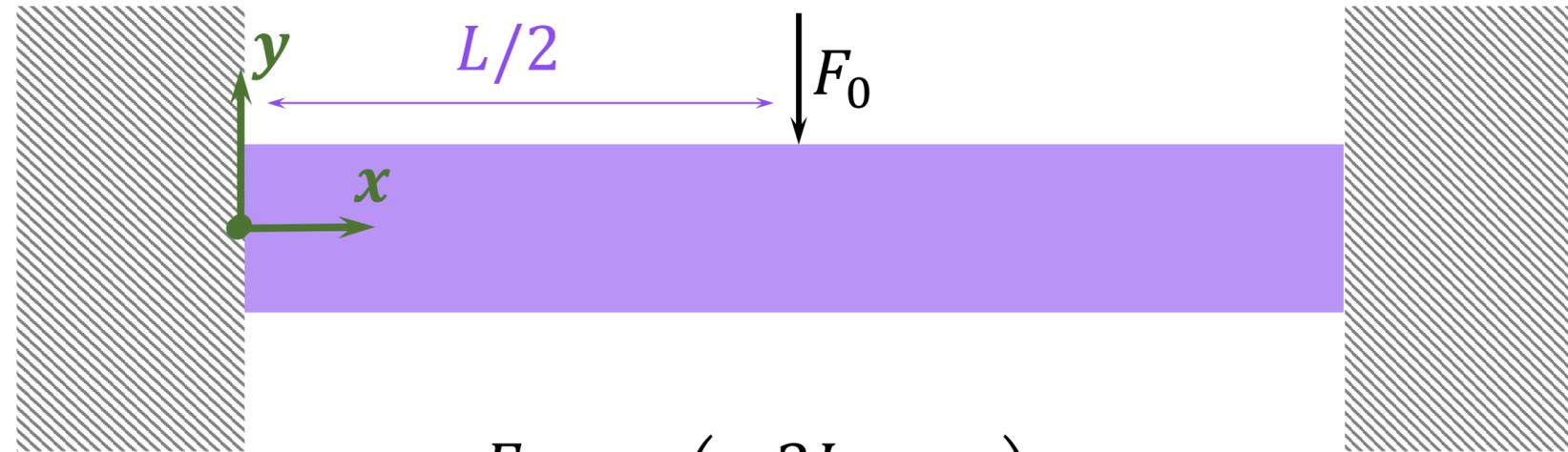
**Nonlinearity**

# EPFL Comparing cantilever and C-C beam



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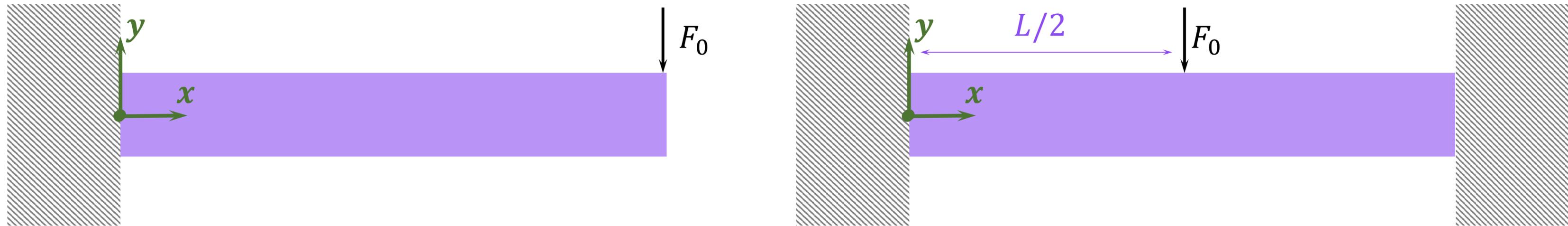
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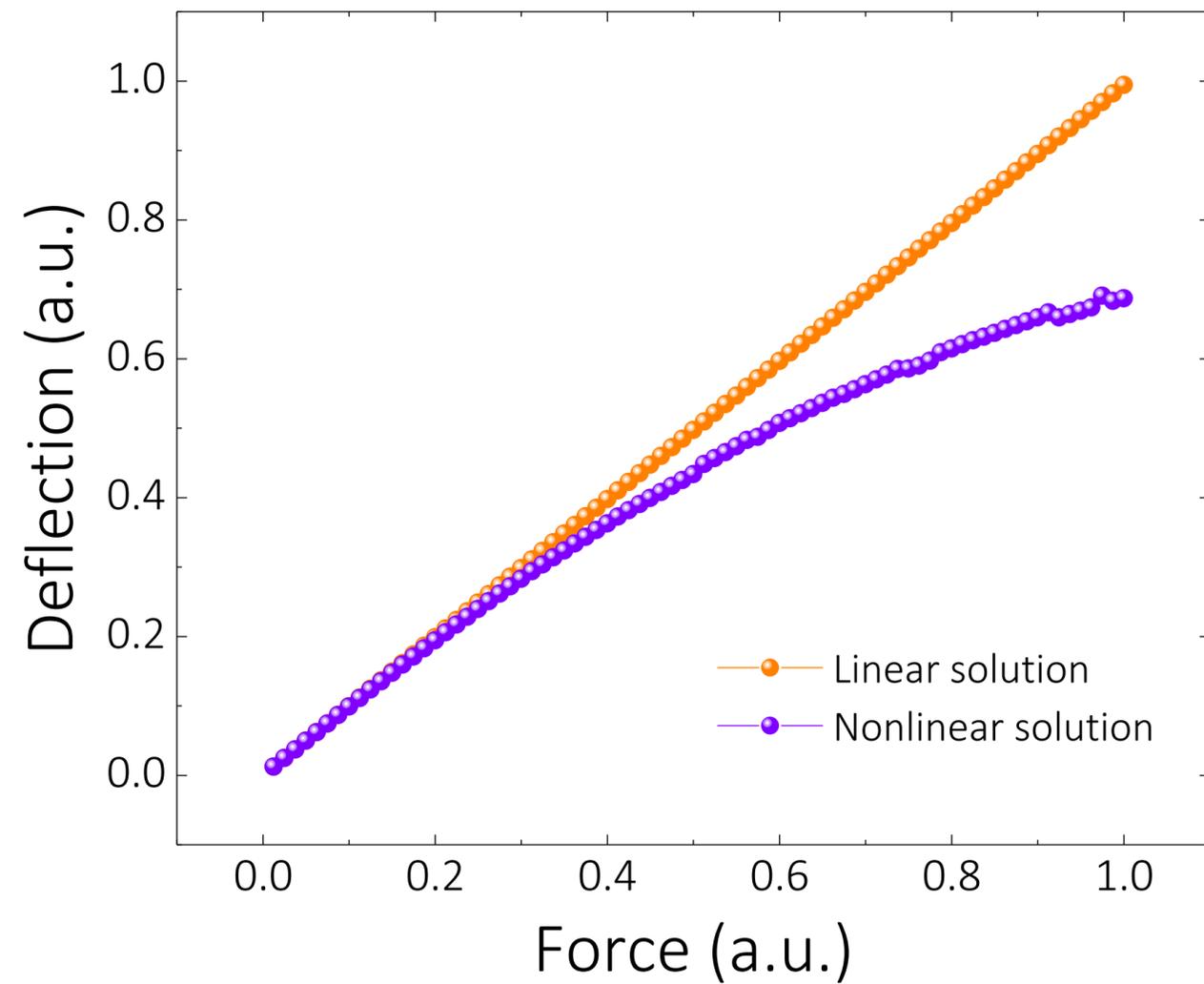
ME426 - Lecture 4 - Force detection

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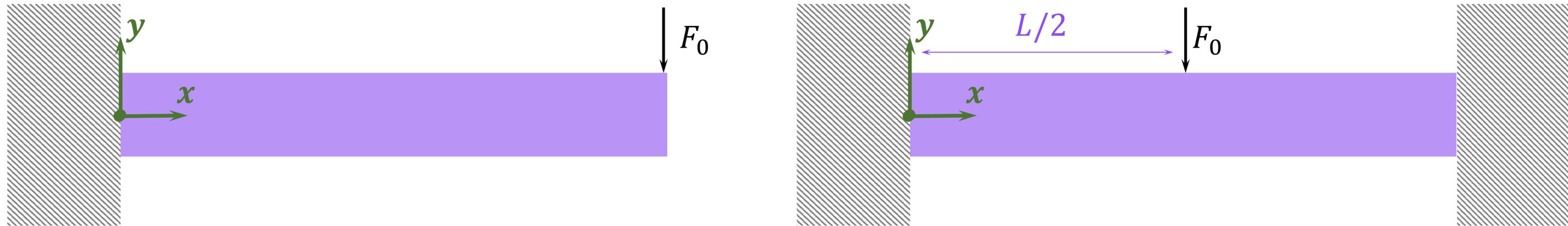
# EPFL Nonlinear effects



ME426 - Lecture 4 - Force detection

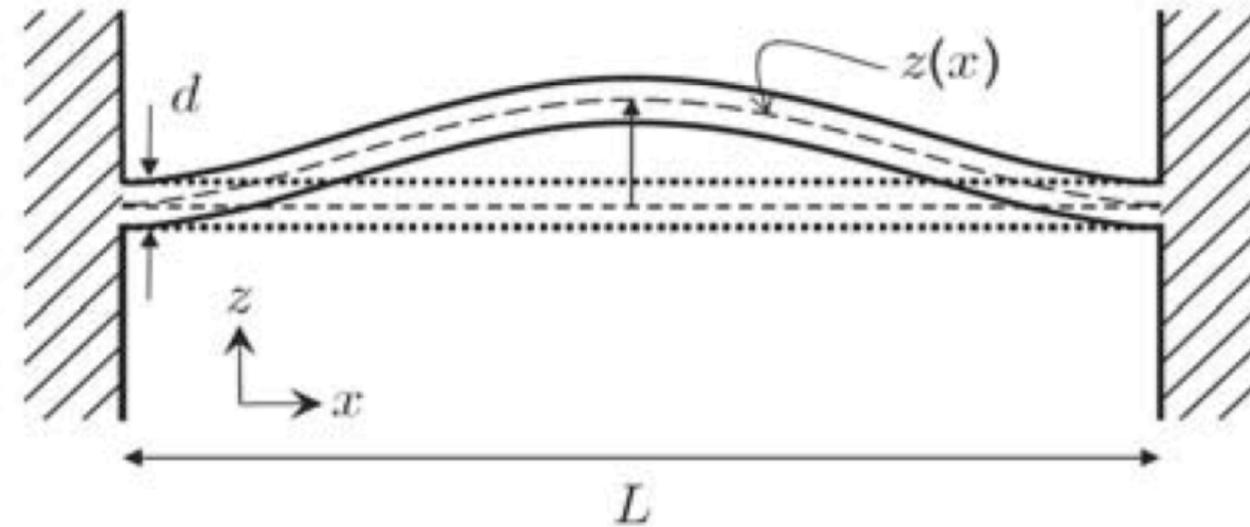


# EPFL Nonlinear effects

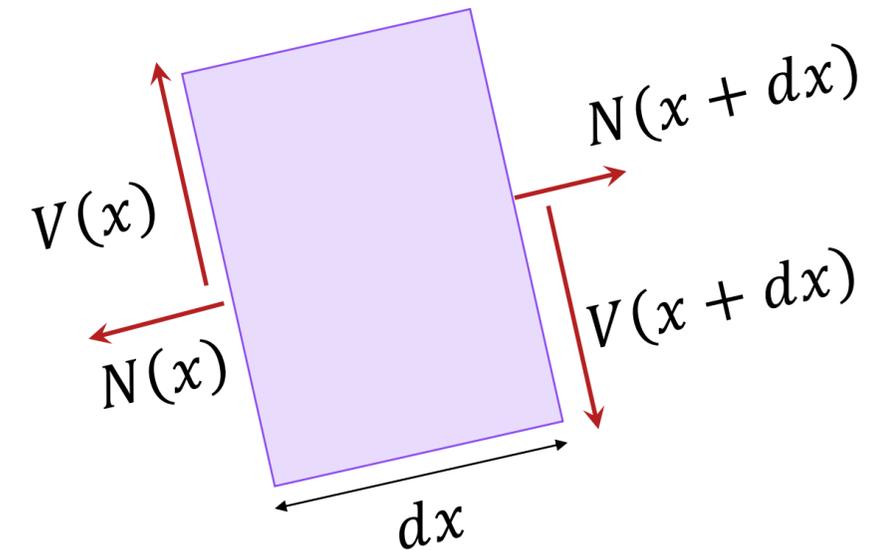


- Nonlinearity happens when the ratio between force and displacement is not constant
- This is caused by additional terms not considered in the description of the deflection
- Cantilever: Small displacements approximation is not valid  $\kappa = \frac{1}{\rho} = \frac{w''(x)}{(1+w'(x)^2)^{3/2}}$ 
  - Nonlinearity happens when deflection is  $w(L) \sim L$
- C-C Beam: Built-in Tension in the beam modifies the stiffness
  - Nonlinearity happens when deflection is  $w(L/2) \sim t$

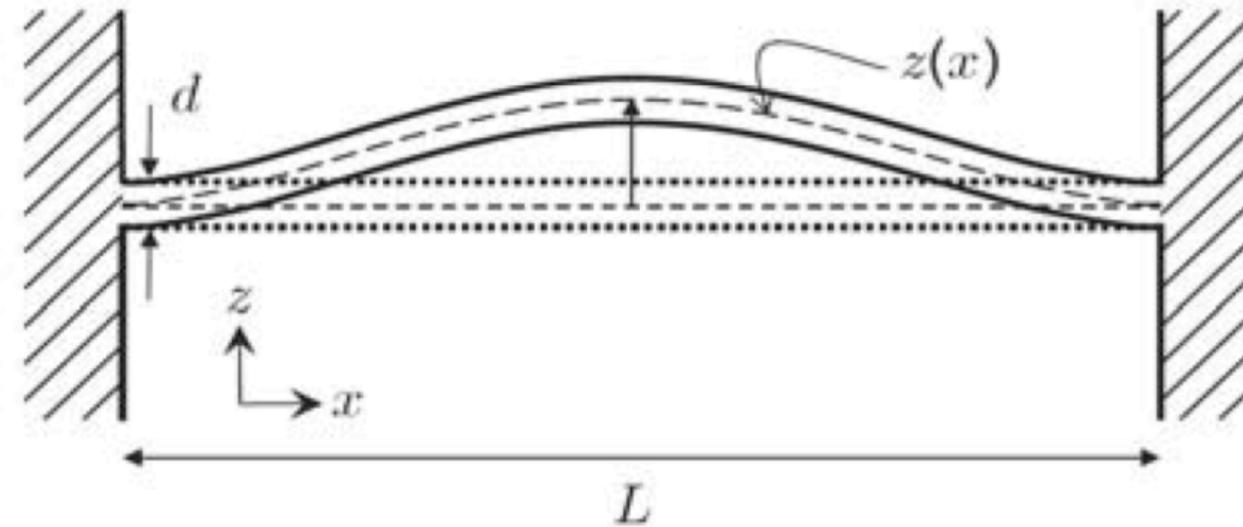
# EPFL Built-in tension



- Deflection of the beam gives rise to an elongation caused by the motion:
  - $\frac{\Delta L}{L} = \varepsilon_x = \frac{1}{2L} \int_0^L w'(x)^2 dx$
- This required elongation causes a stress to appear in the beam
  - $\sigma_x = E\varepsilon_x \rightarrow N = E\varepsilon_x A$

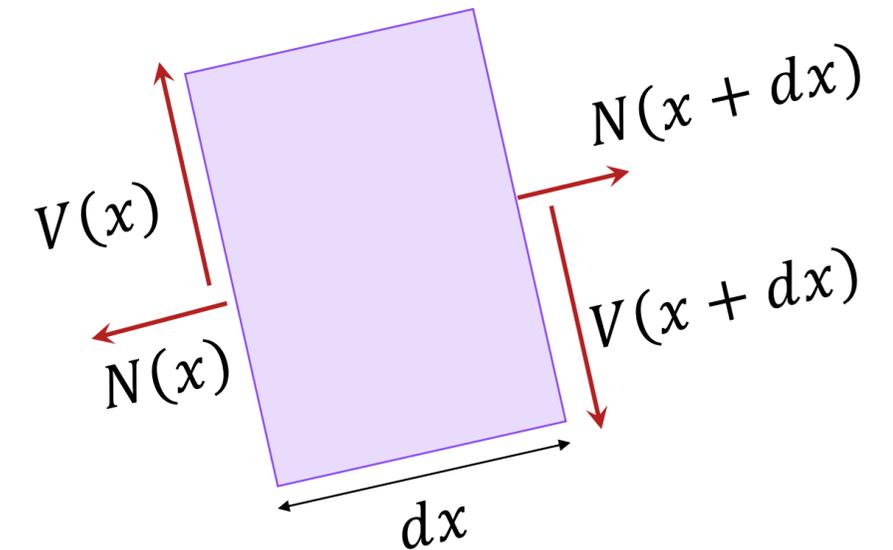


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$$\begin{aligned}
 & -V(x + dx) + V(x) + N \cdot (w'(x + dx) - w'(x)) = \\
 & = -V'(x) dx + Nw''(x)dx = -V'(x) dx + Nw''(x)dx = \\
 & = -V'(x) dx + \frac{EA}{2L} \int_0^L w'(x)^2 dx w''(x)dx
 \end{aligned}$$

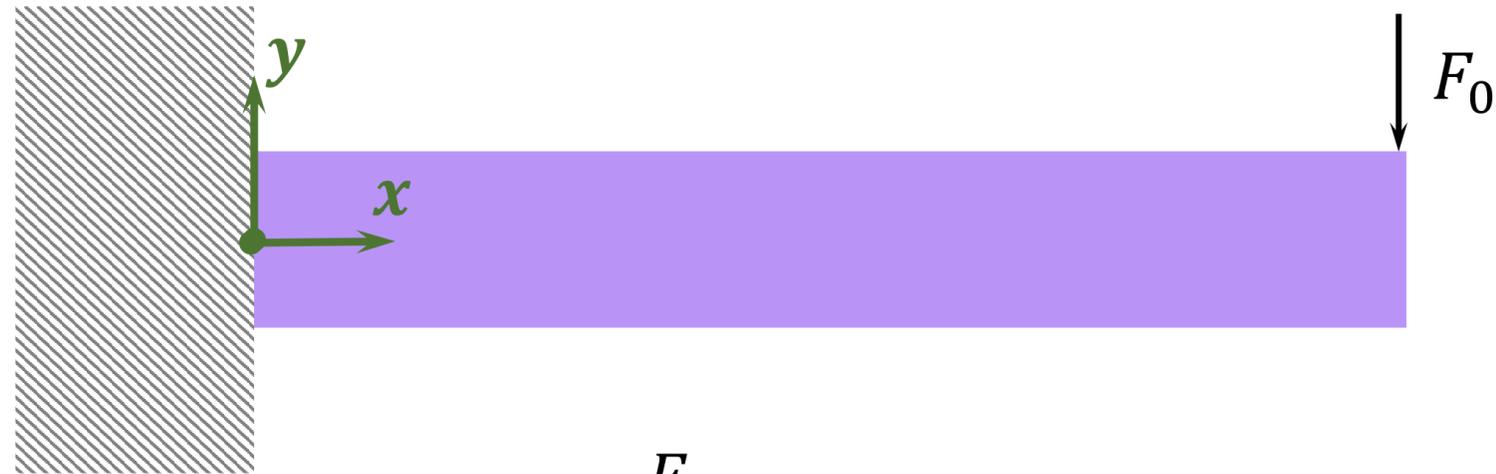


**EPFL**



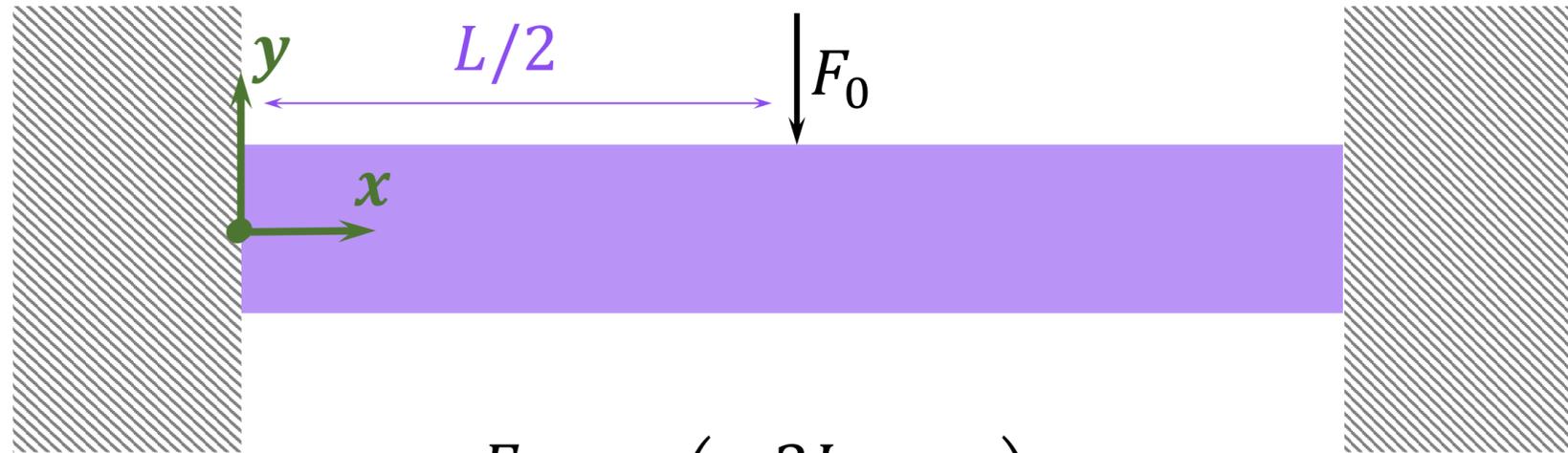
# Composite beams

# EPFL Comparing cantilever and C-C beam



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$$w(L) = -\frac{F_0}{3EI} L^3$$



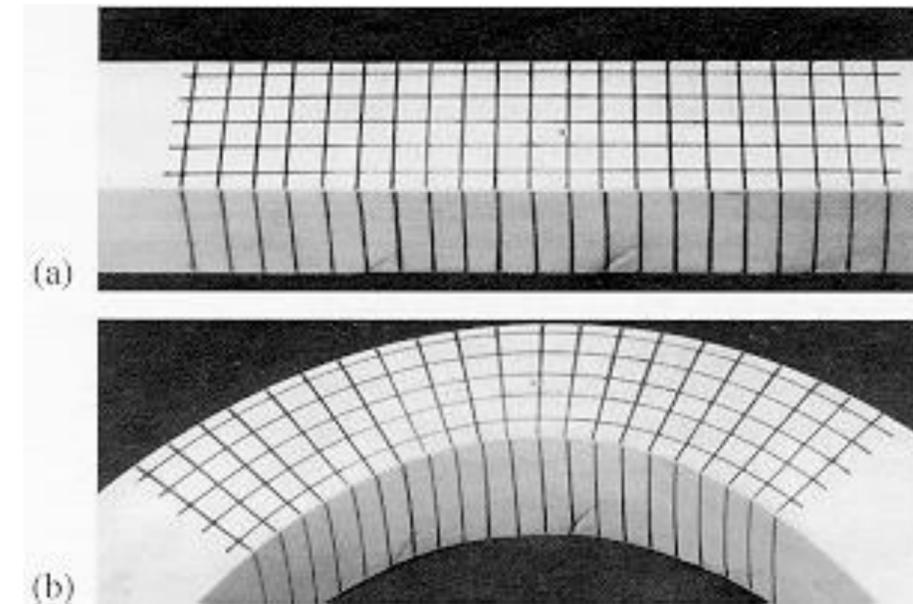
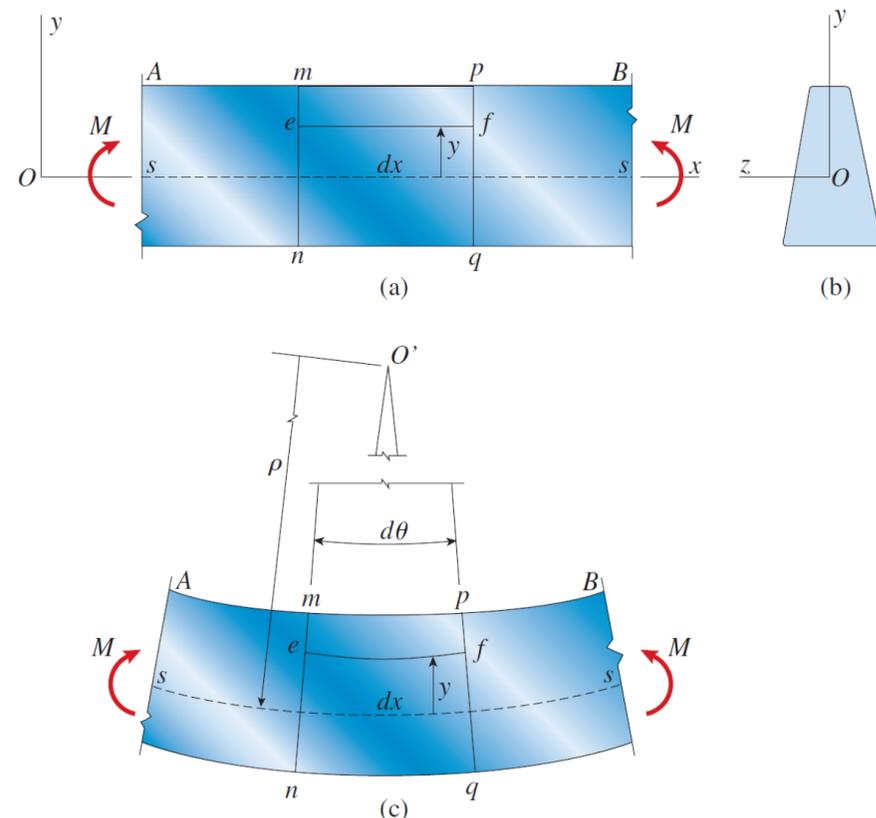
$$w(x) = \frac{F_0}{24EI} x^2 \left( -\frac{3L}{2} + 2x \right) \quad \text{if } x \leq L/2$$

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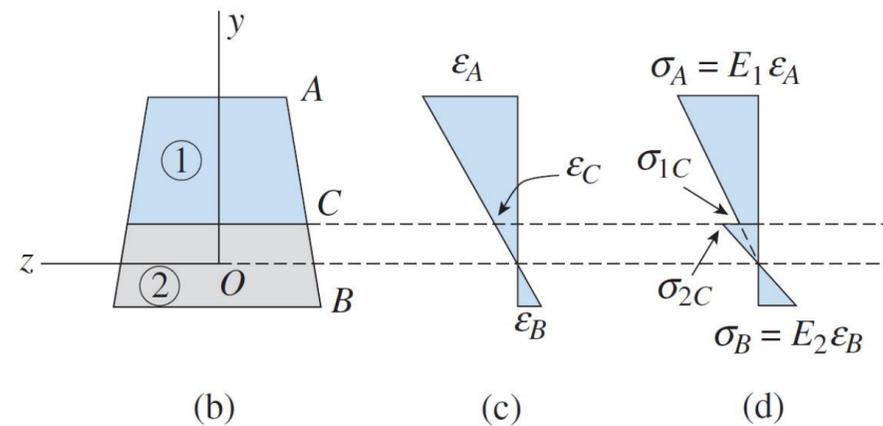
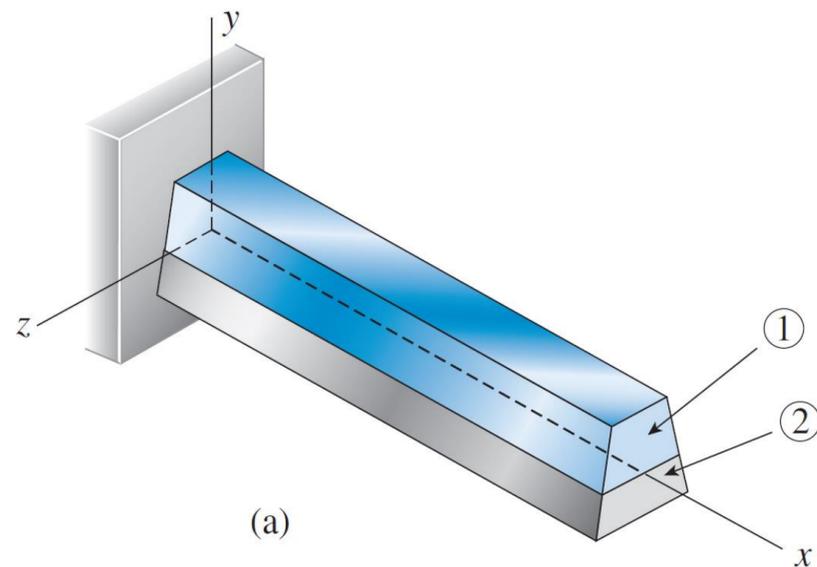
# EPFL Normal Strain in Composite Beams

- When a beam is bending, we can locally define a radius of curvature  $\rho$  and a curvature  $\kappa$
- Planes normal to the beam axis before bending will remain plane after bending
- Part of the beam will expand while part of the beam will contract
- The axis where there is no elongation is called Neutral Axis



# EPFL Normal Strain in Composite Beams

- For the curvature  $\kappa$  of the beam we get:  $\kappa = \frac{d\theta}{ds} = \frac{1}{\rho}$
- After deformation, the neutral axis remains the same
- Any other parallel line:
- $L_f = (\rho - (y - y_0)) d\theta = ds_0 - (y - y_0) \frac{ds_0}{\rho} \rightarrow \epsilon_x = -\frac{y - y_0}{\rho}$



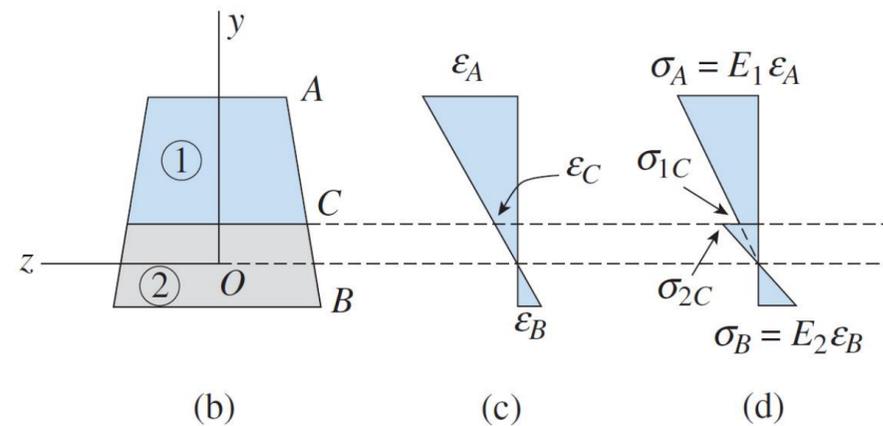
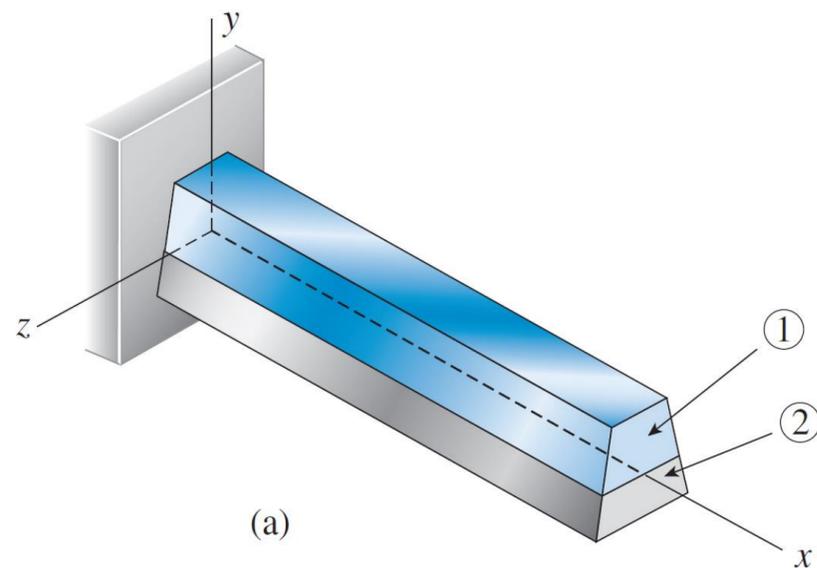
**This is the same for a composite or a monomaterial beam**

# EPFL Normal Stress in Composite Beams

- $\epsilon_x = -\frac{y-y_0}{\rho}$
- What changes is the calculation of stress
- Using Hooke's law:  ~~$\sigma_x = E\epsilon_x = -E\frac{y-y_0}{\rho}$~~

$$\sigma_{x,i} = E_i \epsilon_x = -E_i \frac{y - y_0}{\rho}$$

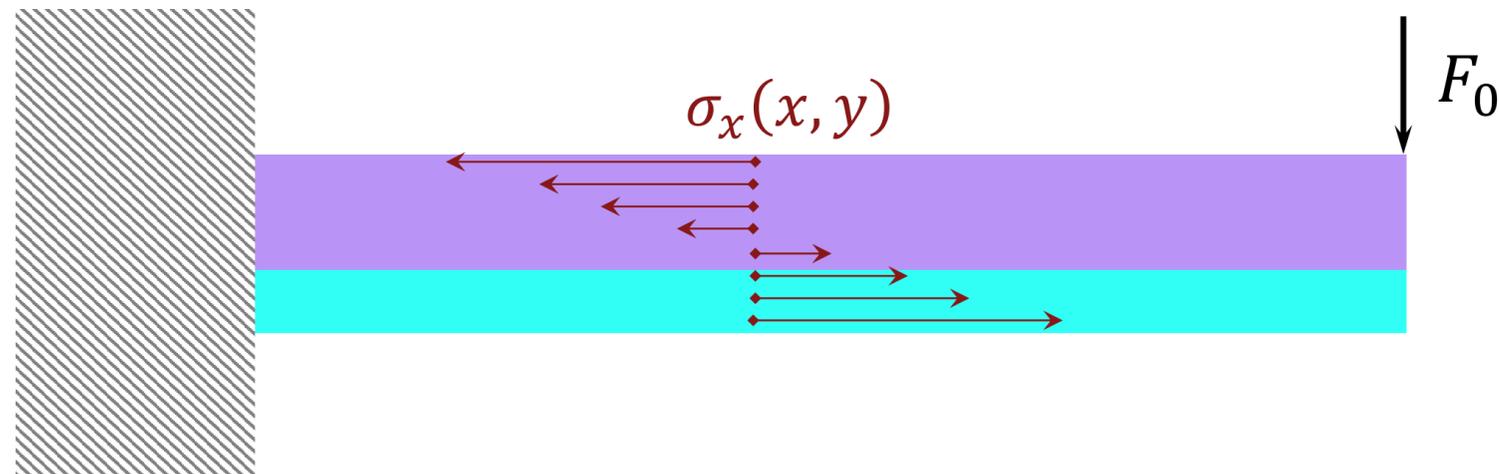
$$\sigma_x(y) = -E(y) \frac{y - y_0}{\rho}$$



# EPFL Neutral axis

- We “cut” and we apply equilibrium:

$$\sum F_x = N = 0$$



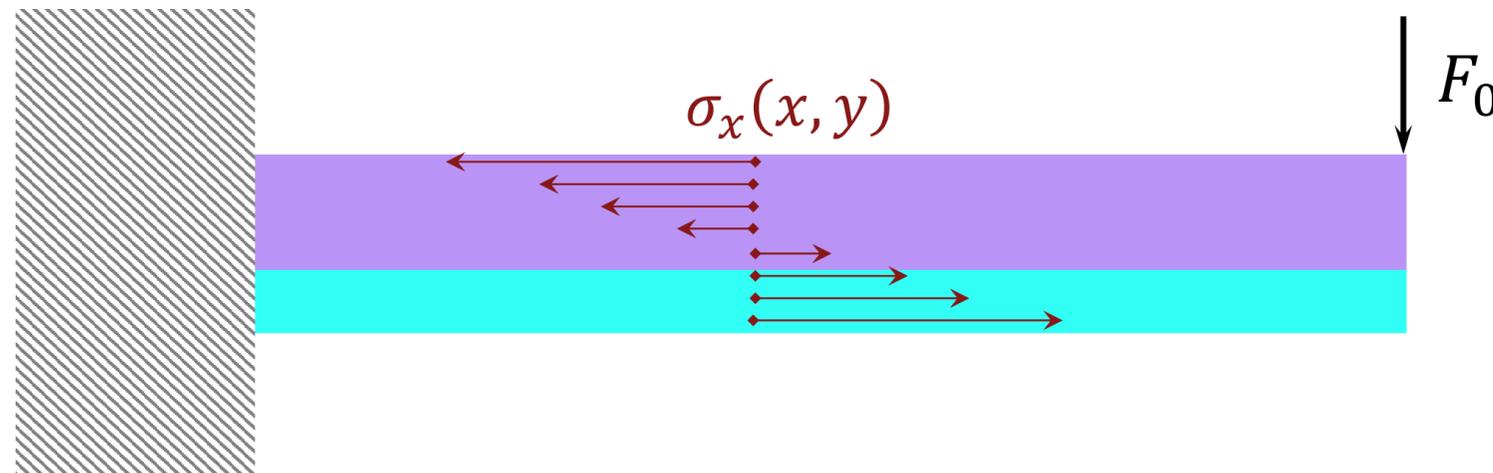
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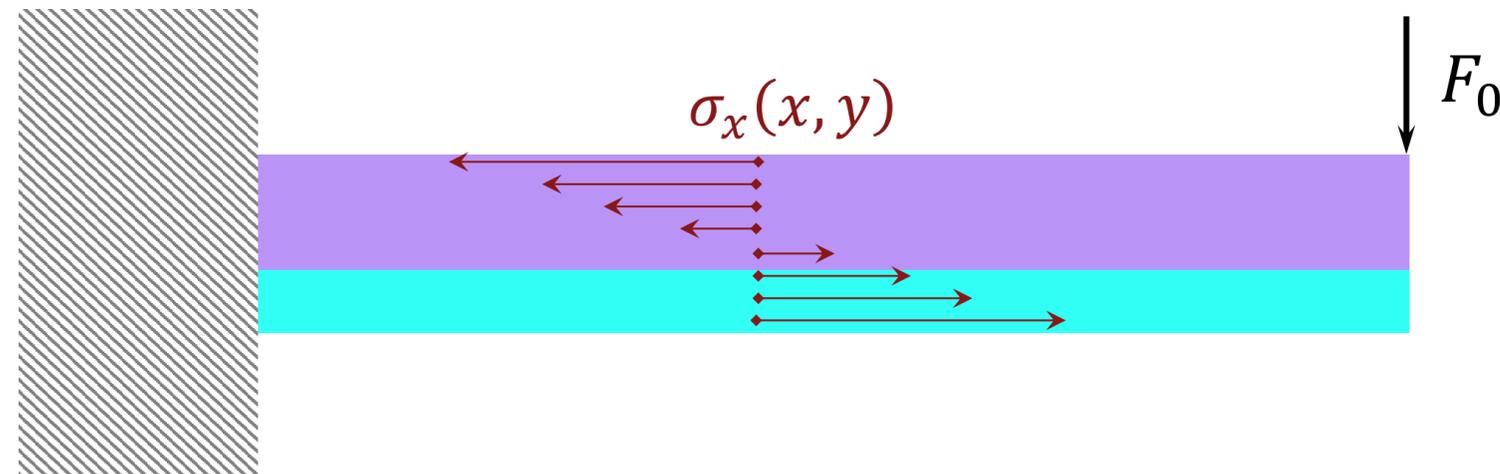
$$N = \iint \sigma_x(x, y) dy dz = \sum_i \iint E_i \varepsilon_x(x, z) dy dz = - \sum_i \iint E_i \frac{(y - y_0)}{\rho} dy dz$$

$$y_0 = \frac{\sum_i \int E_i \frac{y}{\rho} dA}{\sum_i \int \frac{E_i}{\rho} dA} = \frac{\sum_i \int E_i y dA}{\sum_i \int E_i dA} = \frac{\sum_i E_i \int y dA}{\sum_i E_i \int dA} \rightarrow \textit{not the centroid}$$



# EPFL Bending Moment

- We can calculate the moment created by the normal stresses with respect to the neutral axis



# EPFL Bending Moment

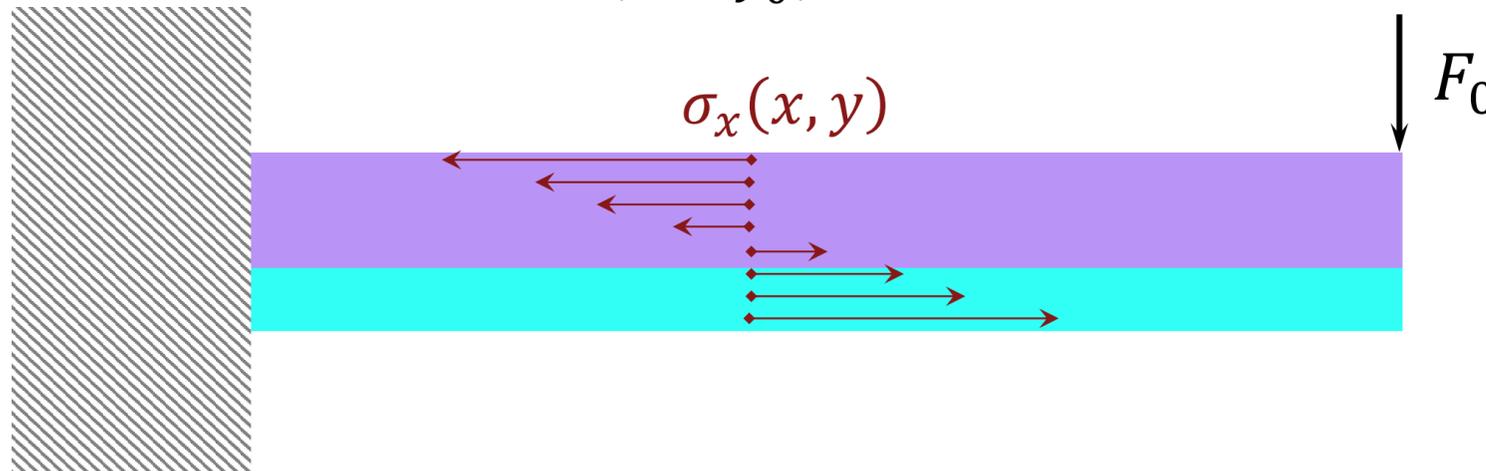
- We can calculate the moment created by the normal stresses with respect to the neutral axis

$$M_z(x) = - \iint \sigma_x(x, y)(y - y_0) dA = \sum_i \iint E_i \frac{(y - y_0)^2}{\rho} dA$$

Bending Stiffness

$$M_z(x) = \frac{1}{\rho} \langle EI_{z,y_0} \rangle; \text{ where } \langle EI_{z,y_0} \rangle = \sum_i E_i \iint (y - y_0)^2 dA = \sum_i E_i I_{z,y_0,i}$$

$$\sigma_x(x, y) = - \frac{E(y)M_z(x)}{\langle EI_{z,y_0} \rangle} (y - y_0) \quad \text{Flexure formula for composite beams}$$

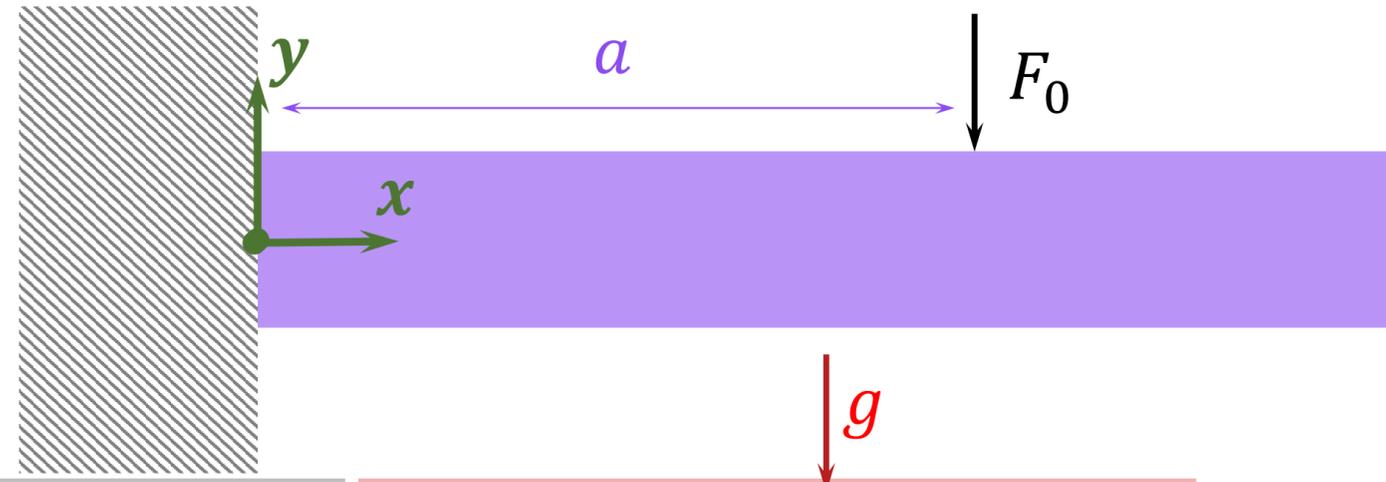


**EPFL**



# Accelerometers

# EPFL Let's put some numbers

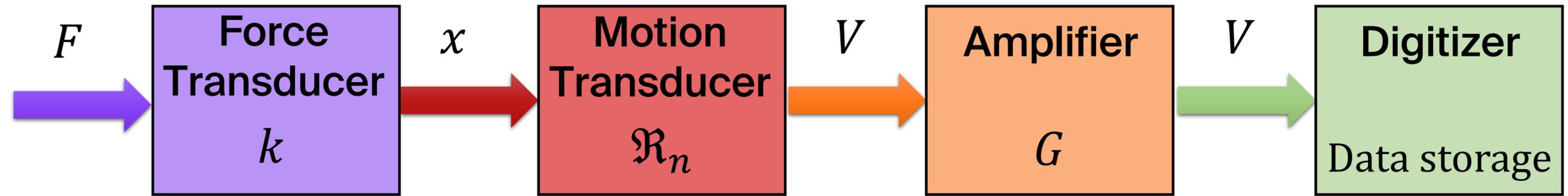


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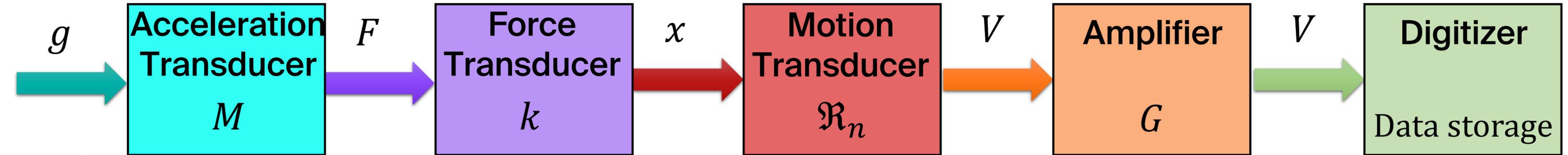
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Tuning fork	$0.1 \times 10^{-3} \times 10^{-3}$	$3 \cdot 10^{-5}$	$2 \cdot 10^{-3}$
Microcantilever	$(500 \times 50 \times 5) \cdot 10^{-6}$	$10^{-9}$	$4 \cdot 10^{-9}$
Nanocantilever	$(10 \times 0.5 \times 0.1) \cdot 10^{-6}$	$3 \cdot 10^{-13}$	$10^{-14}$

In micro/nano scale, gravity and weight can be neglected – but maybe important for vibrations

# EPFL Force detection



# EPFL Acceleration detection



# EPFL Design example

- Large mass
- Soft spring (supporting beam)
- Small gaps

